

TESS Data Release Notes: Sector 60, DR85

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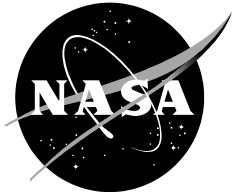
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Acknowledgements

These Data Release Notes provide information on the processing and export of data from the Transiting Exoplanet Survey Satellite (TESS). The data products included in this data release are full frame images (FFIs), target pixel files, light curve files, collateral pixel files, cotrending basis vectors (CBVs), and Data Validation (DV) reports, time series, and associated xml files.

These data products were generated by the TESS Science Processing Operations Center (SPOC, [Jenkins et al., 2016](#)) at NASA Ames Research Center from data collected by the TESS instrument, which is managed by the TESS Payload Operations Center (POC) at Massachusetts Institute of Technology (MIT). The format and content of these data products are documented in the [Science Data Products Description Document \(SDPDD\)](#)¹. The SPOC science algorithms are based heavily on those of the Kepler Mission science pipeline, and are described in the Kepler Data Processing Handbook ([Jenkins, 2020](#)).² The Data Validation algorithms are documented in [Twicken et al. \(2018\)](#) and [Li et al. \(2019\)](#). The [TESS Instrument Handbook](#) ([Vanderspek et al., 2018](#)) contains more information about the TESS instrument design, detector layout, data properties, and mission operations.

The TESS Mission is funded by NASA's Science Mission Directorate.

This report is available in electronic form at
<https://archive.stsci.edu/tess/>

¹<https://archive.stsci.edu/missions/tess/doc/EXP-TESS-ARC-ICD-TM-0014-Rev-F.pdf>

²<https://archive.stsci.edu/kepler/manuals/KSCI-19081-003-KDPH.pdf>

1 Observations

TESS Sector 60 observations include physical orbits 127 and 128 of the spacecraft around the Earth. Data collection was paused for a total of 0.63 days to download data. Data were downloaded approximately every seven days, once near apogee of each orbit and once near perigee.

At TJD 2949.99, the spacecraft entered into safe mode, resulting in instrument turn-off until TJD 2955.04. There are no FFIs, target pixel file data, or light curve data during this time. During these times (as well as when the spacecraft is transmitting data to ground systems) the data and timestamps in target-pixel files and light curve files are set to NULL.

One FFI from Camera 1 in orbit 127 was corrupted prior to downlink from the spacecraft and could not be recovered. The FFIINDEX number is 00738895 (timestamp 2022358084201), and no FFI file for Camera 1 was produced for this cadence.

In total, there are 20.01 days of science data collected in Sector 60.

Table 1: Sector 60 Observation times

	UTC	TJD ^a	Cadence #
Orbit 127a start	2022-12-23 09:30:41	2936.89779	1230799
Orbit 127a end	2022-12-30 04:52:41	2943.70473	1235700
Orbit 127b start	2022-12-30 09:54:41	2943.91446	1235851
Orbit 127b end	2023-01-05 06:22:40	2949.76723	1240065
Orbit 128a start	2023-01-05 11:24:40	2949.97695	1240216
Safe Mode Start	2023-01-05 11:49:07	2949.99325	1240228
Safe Mode End	2023-01-10 12:54:07	2955.03838	1243860
Orbit 128a end	2023-01-11 17:46:40	2956.24223	1244727
Orbit 128b start	2023-01-11 22:50:40	2956.45334	1244879
Orbit 128b end	2023-01-18 01:56:40	2962.58250	1249292

^a TJD = TESS JD = JD - 2,457,000.0

^bThe horizontal black lines mark gaps in the light curves for data downlink.

The spacecraft was pointing at RA (J2000): 125.6742°; Dec (J2000): 77.3113°; Roll: 337.3244°. See the TESS project [Sector 60 observation page](#)³ for the coordinates of the spacecraft pointing and center field-of-view of each camera. Fields-of-view for each camera can be found at the TESS Guest Investigator Office [observations status page](#).⁴ The detailed target list for both 2-minute and 20-second data, as well as the Guest Investigator target lists, can be found at the [Sector 60 observation page](#) and the [observations status page](#).

1.1 Notes on Individual Targets

There are no issues with missing light curves in the 20-second data products. There were 2367 targets chosen for 20-second cadence observations, including 400 PPA stars. TIC 441804565,

³<https://tess.mit.edu/observations/sector-60>

⁴<https://heasarc.gsfc.nasa.gov/docs/tess/sector.html>

341873045, and 1551711135 are blended with comparably bright stars—the contaminating flux for these objects is very large, and the resulting photometry is expected to be unreliable.

For the 2-minute cadence data, 13000 targets were selected. One bright star ($T_{\text{mag}} \lesssim 1.8$) with a large pixel stamp was not processed in the photometric pipeline. The target pixel file with original and calibrated pixel data is provided, but no light curve was produced. Note that the TPF files do not include a background correction for stars without light curves. The affected TIC ID is 229540730.

Three target stars (441804565, 341873045, and 1551711135) are blended with comparably bright stars—the contaminating flux for these objects is very large, and the resulting photometry is expected to be unreliable.

One target (269664165) has a single pixel disjoint from the rest of the optimal aperture, caused by crowding from nearby stars. The target is bright ($T_{\text{mag}} \sim 4.38$) and the disjoint pixel has almost no affect on the light curve, so the photometry of this source is expected to be reliable.

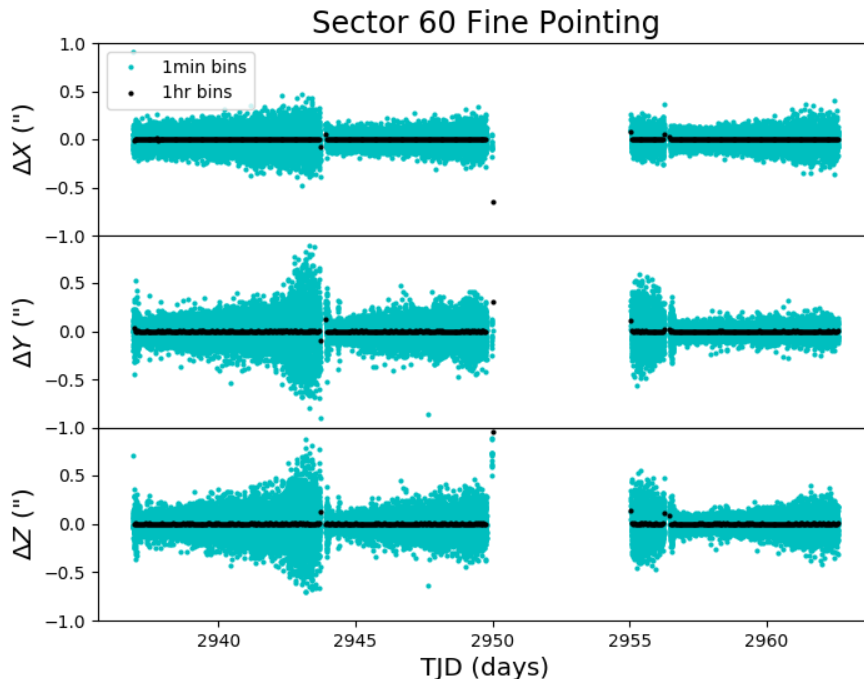


Figure 1: The delta-quaternions from each camera have been converted to spacecraft frame, binned to 1 minute and 1 hour. Long-term trends (such as those caused by differential velocity aberration) have also been removed. The $\Delta X/\Delta Y$ directions represent offsets along the detectors’ rows/columns, while the ΔZ direction represents spacecraft roll.

1.2 Spacecraft Pointing and Momentum dumps

The pointing in Sector 60 was set at +55.5 degrees in ecliptic latitude. Camera 4 alone was used for guiding in the first half of orbit 127. Camera 1 and Camera 4 were both

used for guiding at all other times. One momentum dump was performed at the end of each data downlink, just before data collection resumed. Figure 1 summarizes the pointing performance over the course of the sector based on Fine Pointing telemetry.

1.3 Scattered Light

Figure 2 shows the median value of the background estimate for all targets on a given CCD as a function of time. Figure 3 shows the angle between each camera’s boresight and the Earth or Moon—this figure can be used to identify periods affected by scattered light and the relative contributions of the Earth and Moon to the image backgrounds.

In Sector 60, the Earth introduces scattered light signals at the very start of each orbit.

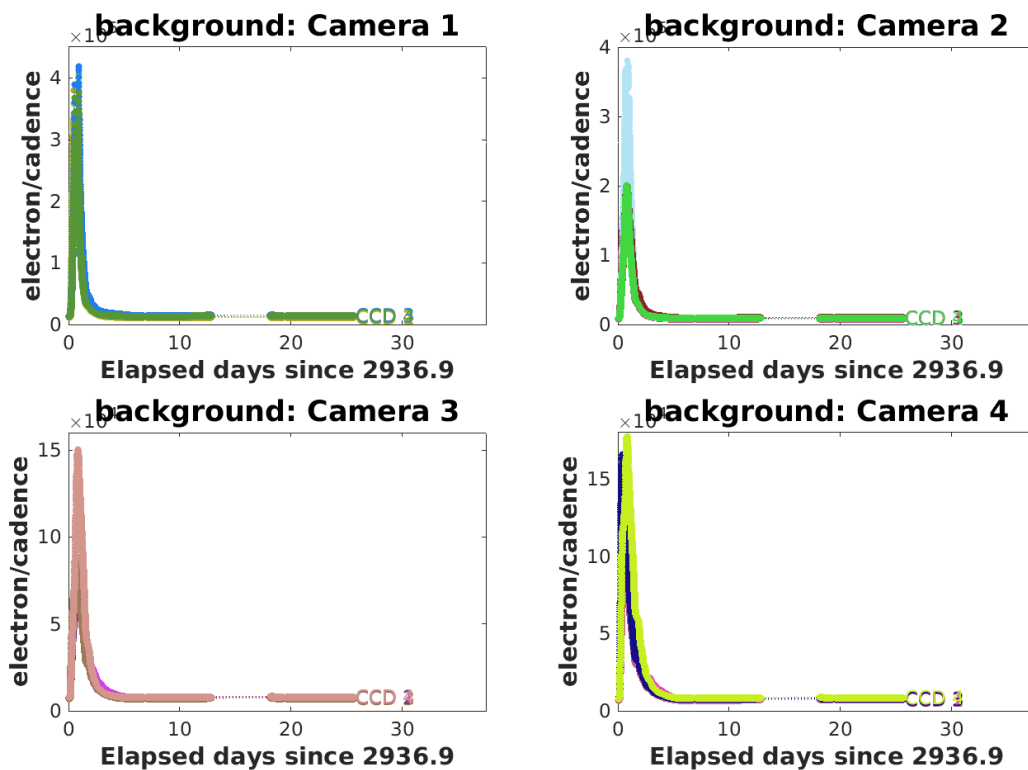


Figure 2: Median background flux across all targets on a given CCD in each camera. The changes are caused by variations in the orientation and distance of the Earth and Moon.

2 Data Anomaly Flags

See the [SDPDD](#) (§9) for a list of data quality flags and the associated binary values used for TESS data, and the [TESS Instrument Handbook](#) for a more detailed description of each flag.

The following flags were not used in Sector 60: bits 1 and 9 (Attitude Tweak and Discontinuity).

Earth/Moon angles for Sector 60 (O127 + O128), Ecl. Dec. = +55.5

2022-12-23 14:42:12 to 2023-01-18 08:34:26 UTC

FFI Cadence Number

246500 247000 247500 248000 248500 249000 249500

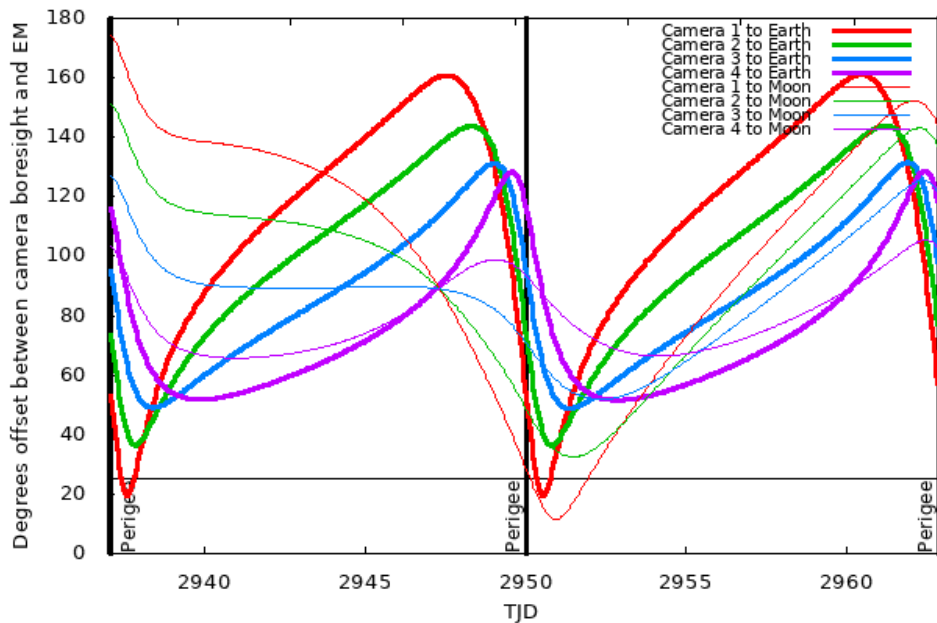


Figure 3: Angle between the four camera boresights and the Earth/Moon as a function of time. When the Earth is within $\sim 25^\circ$ of a camera’s boresight, transiting planet searches may be compromised by high levels of scattered light. At larger angles, up to $\sim 35^\circ$, scattered light patterns and complicated structures may be visible. At yet larger angles, low level patchy features may be visible. Scattered light from the Moon is generally only noticeable below $\sim 35^\circ$. This figure can be used to identify periods affected by scattered light and the relative contributions of the Earth and Moon to the background. However, the background intensity and locations of scattered light features depend on additional factors, such as the Earth/Moon azimuth and distance from the spacecraft.

Cadences marked with bits 2, 3, 4, 6, and 12 (Safe Mode, Coarse Point, Earth Point, Reaction Wheel Desaturation Event, and Straylight) were marked based on spacecraft telemetry. Cadences marked with bits 2 and 4 (Safe Mode and Earth Point) have timestamps and data marked with NULL in the target-pixel files and light curve files.

Note that just after the data downlinks near apogee (midpoint of each orbit), the spacecraft pointing is still settling for 20 to 60 seconds after data collection resumes. Cadences during this time are marked with bit 8 (Manual Exclude, described below).

Cadences marked with bit 5 and 10 (Argabrightening Events and Impulsive Outlier) were identified by the SPOC pipeline. Bit 5 marks a sudden change in the background measurements. In practice, bit 5 flags are caused by rapidly changing glints and unstable pointing at times near momentum dumps. Bit 10 marks an outlier identified by PDC and omitted from the cotrending procedure.

The 20-second data mode includes cadences marked with bit 7 and 11 (Cosmic Ray in Optimal Aperture and Cosmic Ray in Collateral Pixel). These flags indicate cadences affected by cosmic rays that are removed by the pipeline, and can be found in both the TPF

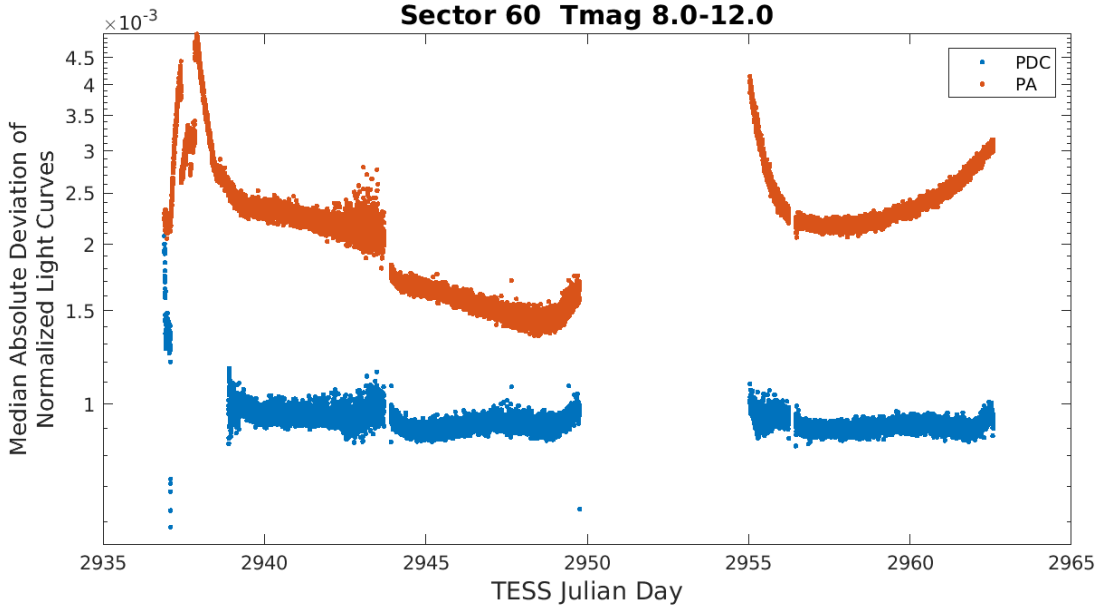


Figure 4: Median absolute deviation (MAD) for the two-minute cadence data from Sector 60, showing the performance of the cotrending after identifying Manual Exclude data quality flags. The MAD is calculated in each cadence across stars with flux variations less than 1% for both the PA (red) and PDC (blue) light curves, where each light curve is normalized by its median flux value. The scatter in the PA light curves is much higher than that for the PDC light curves, and the outliers in the PA light curves are largely absent from the PDC light curves due to the use of the anomaly flags.

and LC files. The data provided in the archive products are corrected for cosmic rays, and a FITS table extension in the TPF and Collateral Pixel File details the cosmic rays identified and removed by the pipeline at the pixel level.

Cadences marked with bit 8 (Manual Exclude) are ignored by PDC, TPS, and DV for cotrending and transit searches. In Sector 60, these cadences were identified using spacecraft telemetry from the fine pointing system. All cadences with pointing excursions >7 arcsec (0.3 pixel) were flagged for manual exclude. Figure 4 shows an assessment of the performance of the cotrending based on the final set of manual excludes.

The predicted stray light flag (bit 12, value 2048) is marked in the FFIs and flags times when the Earth/Moon are near the camera FOVs and may interfere with guiding or saturate the detectors. We strongly recommend that users inspect the FFI data before removing images marked with bit 12, because this bit is set based on predictions from mission planning and is known to be conservative with respect to the quality of data usable for analysis.

The predicted stray light flag (bit 12) is disabled for the 2-minute and 20-second data products. The scattered light exclude flag (bit 13, value 4096) identifies cadences at which individual targets are affected by scattered light

If the Earth/Moon interference is strong enough to saturate the detector, all targets on a CCD slice will be affected and the data are unusable. Cadences with bad calibrations due to saturation are now explicitly marked with bit 15 (value 16384, “Bad Calibration Exclude”). For some cadences, the majority of targets on a CCD may be flagged for scattered light and

not enough valid data remains to derive cotrending basis vectors in PDC. No systematic error correction can be applied at these times. This situation is identified by bit 16 (value 32768, “Insufficient Targets for Error Correction Exclude”).

FFIs were only marked with bits 3, 6, 8, 12, and 15 (Coarse Point, Reaction Wheel Desaturation Events, Manual Exclude, Straylight, and Bad Calibration Exclude). There are no WCS coordinates for FFIs that coincide with momentum dumps.

3 Anomalous Effects

Smear Correction Issues

The following columns were impacted by bright stars in the science frame, and/or upper buffer rows, and/or lower science frame rows, which bleed into the upper serial register resulting in an overestimated smear correction.

- Camera 3, CCD 1, Column 1211 - Star Polaris
- Camera 3, CCD 1, Column 2029 - Star HD 113591
- Camera 3, CCD 2, Column 1129 - Star HD 112599
- Camera 3, CCD 3, Column 1939 - Star HD 116257
- Camera 3, CCD 4, Column 1381 - Star HD 1006
- Camera 4, CCD 1, Column 365 - Star HD 188522
- Camera 4, CCD 1, Column 1896 - Star HD 170592
- Camera 4, CCD 2, Column 988 - Star HD 157462
- Camera 4, CCD 2, Column 1936 - Star HD 150578

4 Pipeline Performance and Results

4.1 Light Curves and Photometric Precision

Figure 5 gives the PDC goodness metrics for the two-minute cadence data, with residual correlation goodness and introduced noise goodness shown on a scale between 0 (bad) and 1 (good). The performance of PDC is very good and generally uniform over most of the field of view. Figure 6 shows the achieved Combined Differential Photometric Precision (CDPP) at 1-hour timescales for all two-minute targets.

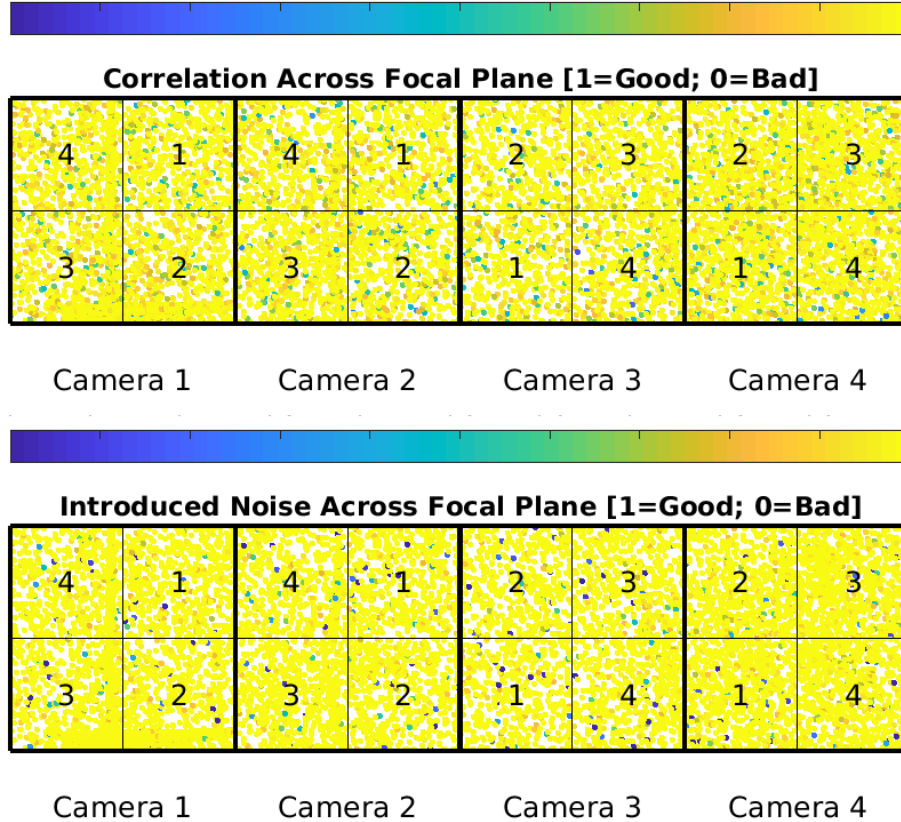


Figure 5: PDC residual correlation goodness metric (top panel) and PDC introduced noise goodness metric (bottom panel) for the two-minute cadence data. The metric values are shown on a focal plane map indicating the camera and CCD location of each target. The correlation goodness metric is calibrated such that a value greater than 0.8 means there is less than 10% mean absolute correlation between the target under study and all other targets on the CCD. The introduced noise metric is calibrated such that a value greater than 0.8 means the power in broad-band introduced noise is below the level of uncertainties in the flux values.

4.2 Transit Search and Data Validation

In Sector 60, the two-minute light curves of 12999 targets were subjected to the transit search in TPS. Of these, Threshold Crossing Events (TCEs) at the 7.1σ level were generated for 769 targets.

We employed an iterative method when conducting the Sector 60 transit search. The top panel of Figure 7 shows the number of TCEs at a given cadence that exhibit a transit signal from an initial run of TPS. The $3\text{-}\sigma$ peaks were used to define de-emphasis weights for a second run of TPS, the results of which are shown in the bottom panel of Figure 7. The final set of TCEs and the results reported here are based on the second run of TPS. The values of the adopted de-emphasis weights are provided in the DV timeseries data products for targets with TCEs.

The top panel of Figure 8 shows the distribution of orbital periods for the final set of TCEs found in Sector 60. The bimodal nature of the period distribution is due to a large

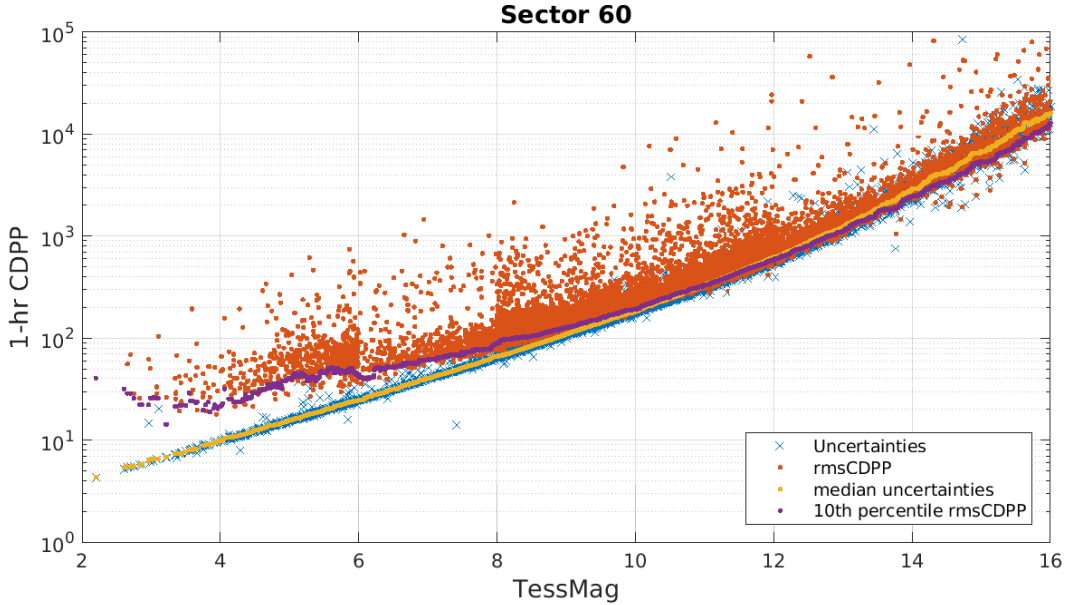


Figure 6: 1-hour CDPP. The red points are the RMS CDPP measurements for the 12999 light curves from Sector 60 plotted as a function of TESS magnitude. The blue x’s are the uncertainties, scaled to 1-hour timescale. The purple curve is a moving 10th percentile of the RMS CDPP measurements, and the gold curve is a moving median of the 1-hr uncertainties.

number of two-transit detections with periods ranging from 8 to 20 days, many of which are likely false positives. Two-transit TCEs can be identified with the *NTRANS* keyword in the headers of the dv-timeseries FITS files. The vertical histogram in the right panel of Figure 8 shows the distribution of transit depths derived from limb-darkened transiting planet model fits for TCEs. The model transit depths range down to the order of 100 ppm, but the bulk of the transit depths are considerably larger.

A search for additional TCEs in potential multiple planet systems was conducted in DV through calls to TPS. A total of 1029 TCEs were ultimately identified in the SPOC pipeline on 769 unique target stars. Table 2 provides a breakdown of the number of TCEs by target. Note that targets with large numbers of TCEs are likely to include false positives.

Table 2: Sector 60 TCE Numbers

Number of TCEs	Number of Targets	Total TCEs
1	555	555
2	182	364
3	22	66
4	7	28
5	2	10
6	1	6
–	769	1029

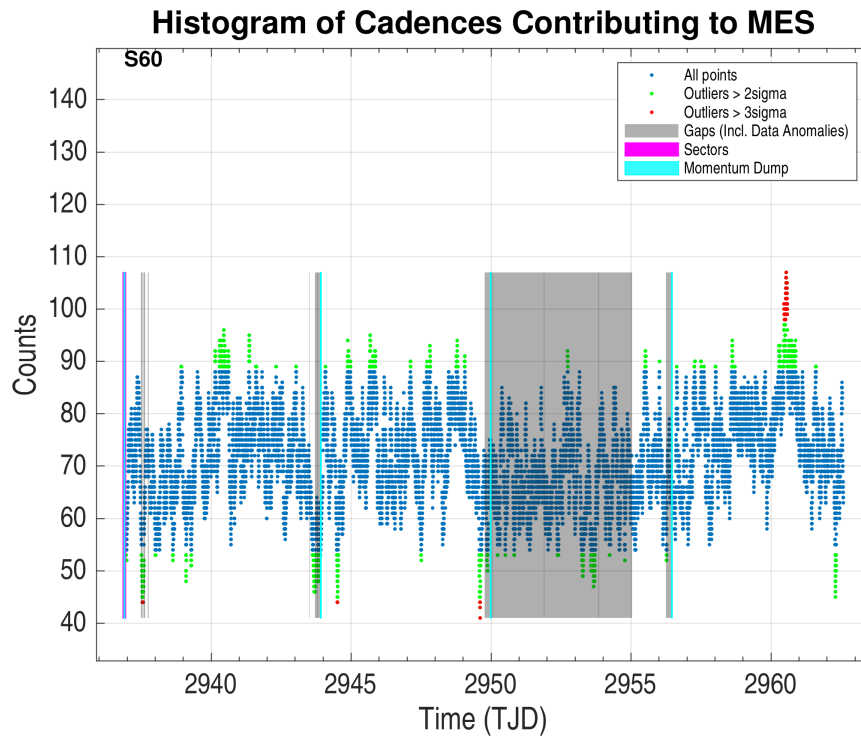
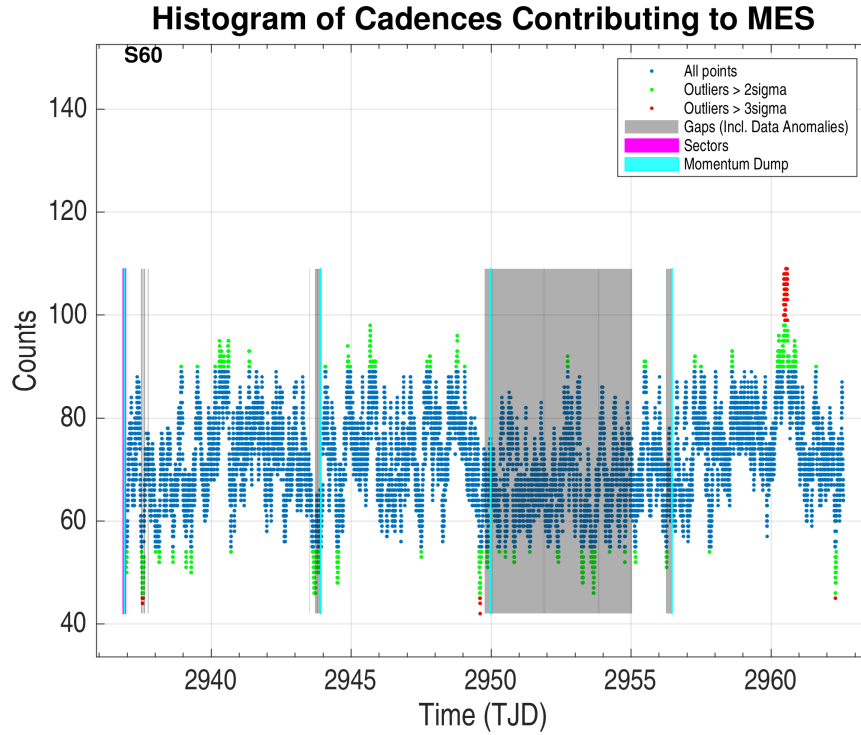


Figure 7: Top panel: Number of TCEs at a given cadence exhibiting a transit signal, based on an initial run of TPS. Any isolated peaks are caused by single events that result in spurious TCEs. These peaks were used to define de-emphasis weights that suppress problematic epochs for the transit detection statistics in a second iteration of TPS. Bottom panel: Results from the second run of TPS.

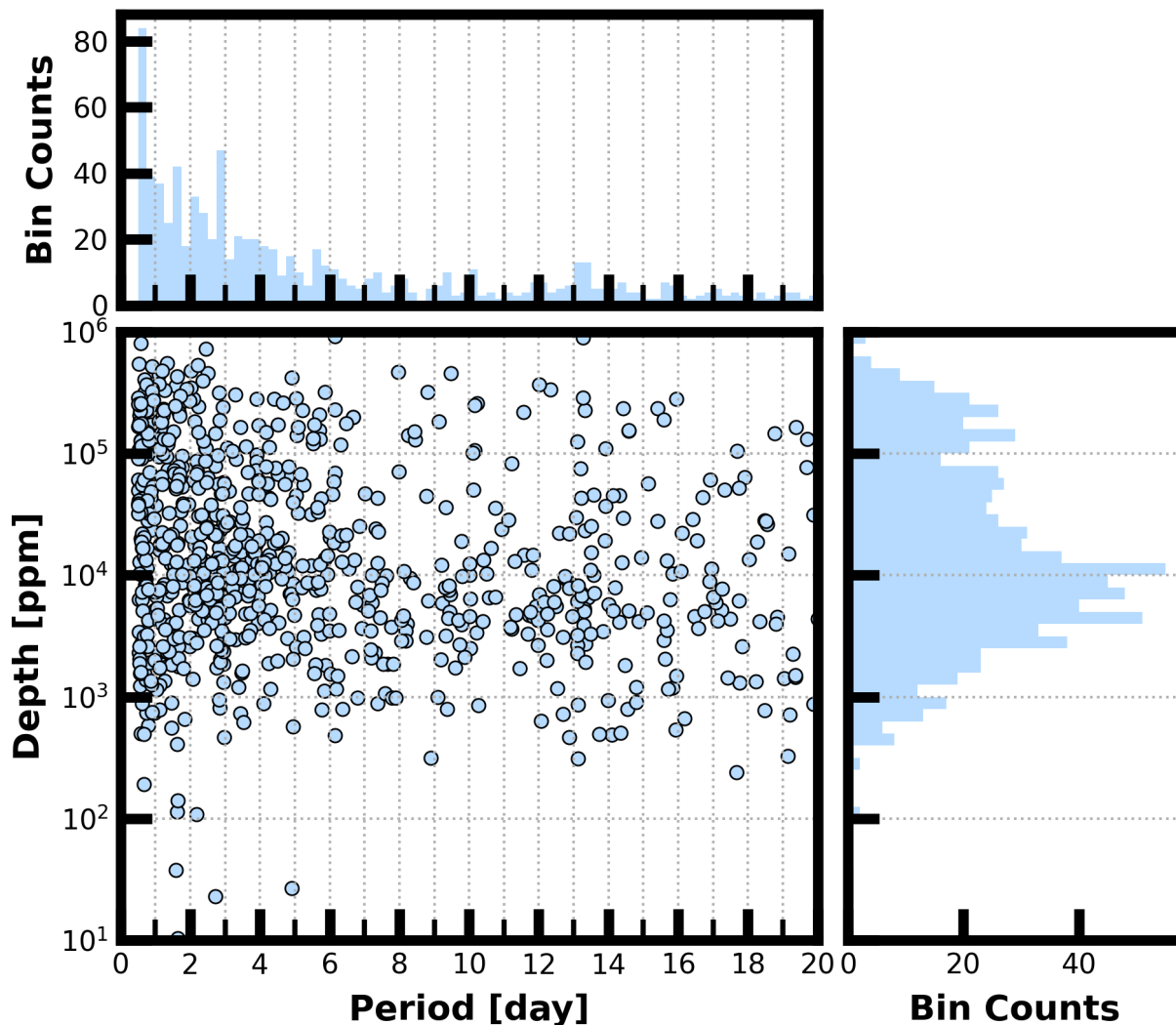


Figure 8: Lower Left Panel: Transit depth as a function of orbital period for the 1029 TCEs identified for the Sector 60 search. For enhanced visibility of long period detections, TCEs with orbital period <0.5 days are not shown. Reported depth comes from the DV limb-darkened transit fit depth when available, and the DV trapezoid model fit depth when not available. Top Panel: Orbital period distribution of the TCEs shown in the lower left panel. Right Panel: Transit depth distribution for the TCEs shown in the lower left panel.

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Acronyms and Abbreviation List

BTJD	Barycentric-corrected TESS Julian Date
CAL	Calibration Pipeline Module
CBV	Cotrending Basis Vector
CCD	Charge Coupled Device
CDPP	Combined Differential Photometric Precision
COA	Compute Optimal Aperture Pipeline Module
CSCI	Computer Software Configuration Item
CTE	Charge Transfer Efficiency
Dec	Declination
DR	Data Release
DV	Data Validation Pipeline Module
DVA	Differential Velocity Aberration
FFI	Full Frame Image
FIN	FFI Index Number
FITS	Flexible Image Transport System
FOV	Field of View
FPG	Focal Plane Geometry model
KDPH	Kepler Data Processing Handbook
KIH	Kepler Instrument Handbook
KOI	Kepler Object of Interest
MAD	Median Absolute Deviation
MAP	Maximum A Posteriori
MAST	Mikulski Archive for Space Telescopes
MES	Multiple Event Statistic
NAS	NASA Advanced Supercomputing Division
PA	Photometric Analysis Pipeline Module

PDC Pre-Search Data Conditioning Pipeline Module

PDC-MAP Pre-Search Data Conditioning Maximum A Posteriori algorithm

PDC-msMAP Pre-Search Data Conditioning Multiscale Maximum A Posteriori algorithm

PDF Portable Document Format

POC Payload Operations Center

POU Propagation of Uncertainties

PPA Photometer Performance Assessment

ppm Parts-per-million

PRF Pixel Response Function

RA Right Ascension

RMS Root Mean Square

SAP Simple Aperture Photometry

SDPDD Science Data Products Description Document

SNR Signal-to-Noise Ratio

SPOC Science Processing Operations Center

SVD Singular Value Decomposition

TCE Threshold Crossing Event

TESS Transiting Exoplanet Survey Satellite

TIC TESS Input Catalog

TIH TESS Instrument Handbook

TJD TESS Julian Date

TOI TESS Object of Interest

TPS Transiting Planet Search Pipeline Module

UTC Coordinated Universal Time

WCS World Coordinate System

XML Extensible Markup Language